



## **DEPARTMENT OF TRANSPORTATION**

### **Federal Aviation Administration**

#### **14 CFR Part 39**

**[Docket No. FAA-2025-5390; Project Identifier MCAI-2025-01470-T; Amendment  
39-23220; AD 2025-25-12]**

**RIN 2120-AA64**

#### **Airworthiness Directives; Airbus SAS Airplanes**

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Final rule; request for comments.

**SUMMARY:** The FAA is superseding Airworthiness Directive (AD) 2025-13-12, which applied to all Airbus SAS Model A350-941 and -1041 airplanes. AD 2025-13-12 required replacement of any affected elevator flight control remote module (FCRM), prohibited the installation of aileron or spoiler FCRMs in place of elevator or rudder FCRMs, and limited the installation of FCRMs under certain conditions. Since the FAA issued AD 2025-13-12, the manufacturer developed a modification that eliminates the unsafe condition. This AD continues to require the actions of AD 2025-13-12. This AD also requires installation of the flight control and guidance system (FCGS) primary computer (PRIM) P14.1.3 and secondary computer (SEC) S14.1.2 software standards and prohibits installation of earlier software standards. The FAA is issuing this AD to address the unsafe condition on these products.

**DATES:** This AD is effective [INSERT DATE 15 DAYS AFTER DATE OF  
PUBLICATION IN THE FEDERAL REGISTER].

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of [INSERT DATE 15 DAYS AFTER DATE OF PUBLICATION IN THE FEDERAL REGISTER].

The FAA must receive comments on this AD by [INSERT DATE 45 DAYS AFTER DATE OF PUBLICATION IN THE Federal Register].

**ADDRESSES:** You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

- Federal eRulemaking Portal: Go to [regulations.gov](https://www.regulations.gov). Follow the instructions for submitting comments.

- Fax: (202) 493-2251.

- Mail: U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE, Washington, DC 20590.

- Hand Delivery: Deliver to Mail address above between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

*AD Docket:* You may examine the AD docket at [regulations.gov](https://www.regulations.gov) under Docket No. FAA-2025-5390; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this final rule, the mandatory continuing airworthiness information (MCAI), any comments received, and other information. The street address for Docket Operations is listed above.

*Material Incorporated by Reference:*

- For European Union Aviation Safety Agency (EASA) material identified in this AD, contact EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; telephone +49 221 8999 000; email [ADs@easa.europa.eu](mailto:ADs@easa.europa.eu). You may find this material on the EASA website at [ad.easa.europa.eu](https://ad.easa.europa.eu).

- You may view this material at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206-231-3195. It is also available at [regulations.gov](https://www.regulations.gov) under Docket No. FAA-2025-5390.

**FOR FURTHER INFORMATION CONTACT:** Frank Carreras, Aviation Safety Engineer, FAA, 2200 South 216th St., Des Moines, WA 98198; phone: 206-231-3539; email: Frank.Carreras@faa.gov.

**SUPPLEMENTARY INFORMATION:**

**Comments Invited**

The FAA invites you to send any written data, views, or arguments about this final rule. Send your comments using a method listed under the ADDRESSES section. Include “Docket No. FAA-2025-5390; Project Identifier MCAI-2025-01470-T” at the beginning of your comments. The most helpful comments reference a specific portion of the final rule, explain the reason for any recommended change, and include supporting data. The FAA will consider all comments received by the closing date and may amend this final rule because of those comments.

Except for Confidential Business Information (CBI) as described in the following paragraph, and other information as described in 14 CFR 11.35, the FAA will post all comments received, without change, to regulations.gov, including any personal information you provide. The agency will also post a report summarizing each substantive verbal contact received about this final rule.

**Confidential Business Information**

CBI is commercial or financial information that is both customarily and actually treated as private by its owner. Under the Freedom of Information Act (FOIA) (5 U.S.C. 552), CBI is exempt from public disclosure. If your comments responsive to this AD contain commercial or financial information that is customarily treated as private, that you actually treat as private, and that is relevant or responsive to this AD, it is important that you clearly designate the submitted comments as CBI. Please mark each page of your submission containing CBI as “PROPIN.” The FAA will treat such marked submissions as confidential under the FOIA, and they will not be placed in the public

docket of this AD. Submissions containing CBI should be sent to Frank Carreras, Aviation Safety Engineer, FAA, 2200 South 216th St., Des Moines, WA 98198; phone: 206-231-3539; email: Frank.Carreras@faa.gov. Any commentary that the FAA receives which is not specifically designated as CBI will be placed in the public docket for this rulemaking.

### **Background**

The FAA issued AD 2025-13-12, Amendment 39-23078 (90 FR 30577, July 10, 2025) (AD 2025-13-12), for all Model A350-941 and -1041 airplanes. AD 2025-13-12 was prompted by an MCAI originated by EASA, which is the Technical Agent for the Member States of the European Union. EASA issued AD 2025-0129, dated June 5, 2025 (EASA AD 2025-0129), to correct an unsafe condition. EASA AD 2025-0129 states that an occurrence of loss of control of an outboard aileron surface was reported. Subsequent investigations determined that the electronic card of the FCRM of that aileron had been contaminated by hydraulic fluid. In addition, EASA determined that certain servocontrols were exposed to hydraulic contamination before delivery of the airplane to its first operator. Due to the similarity of design, elevator and rudder FCRMs could be subject to the same failure mode. EASA AD 2025-0129 also stated the AD was considered an interim action and that further AD action might follow.

AD 2025-13-12 required replacement of any affected elevator FCRM and prohibited the installation of aileron or spoiler FCRMs in place of elevator or rudder FCRMs. AD 2025-13-12 also limited the installation of aileron or spoiler FCRMs in place of elevator or rudder FCRMs. The FAA issued AD 2025-13-12 to address the unsafe condition on all Model A350-941 and -1041 airplanes.

### **Actions Since AD 2025-13-12 Was Issued**

Since the FAA issued AD 2025-13-12, EASA superseded EASA AD 2025-0129 with EASA AD 2025-0197R1, dated September 26, 2025 (EASA AD 2025-0197R1)

(also referred to as the MCAI), to correct an unsafe condition for all Airbus SAS Model A350-941 and -1041 airplanes. After issuance of EASA AD 2025-0129, Airbus developed a modification, which includes installing FCGS PRIM P14.1.3 and SEC S14.1.2 software standards, to eliminate the unsafe condition.

The preamble to AD 2025-13-12 explained that the FAA considered that AD an interim action. The FAA has determined that further rulemaking is indeed necessary to mandate the terminating action developed by the airplane manufacturer.

The FAA is issuing this AD to address the unsafe condition on these products. You may examine the MCAI in the AD docket at [regulations.gov](https://www.regulations.gov) under Docket No. FAA-2025-5390.

### **Explanation of Retained Requirements**

Although this AD does not explicitly restate the requirements of AD 2025-13-12, this AD retains all of the requirements of AD 2025-13-12. Those requirements are referenced in EASA AD 2025-0197R1, which, in turn, is referenced in paragraph (g) of this AD.

### **Material Incorporated by Reference Under 1 CFR Part 51**

The FAA reviewed EASA AD 2025-0197R1. This material describes procedures for replacing affected FCRMs, which have been exposed to hydraulic fuel contamination, with serviceable FCRMs and repairing any related hydraulic leaks. EASA AD 2025-0197R1 prohibits “swapping” elevator or rudder FCRMs with spoiler or aileron FCRMs and installing an FCRM unless it is a serviceable FCRM. EASA AD 2025-0197R1 also describes procedures for modifying the FCGS by installing PRIM P14.1.3 and SEC S14.1.2 software standards and prohibits installation of earlier FCGS software standards after modification. EASA AD 2025-0197R1 specifies that the modification terminates the replacement of affected FCRMs, the prohibition for swapping elevator or rudder FCRMs with spoiler or aileron FCRMs, and the limitation for installing only a serviceable FCRM.

This material is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the ADDRESSES section.

### **FAA's Determination**

These products have been approved by the civil aviation authority of another country and are approved for operation in the United States. Pursuant to the FAA's bilateral agreement with this State of Design Authority, that authority has notified the FAA of the unsafe condition described in the MCAI referenced above. The FAA is issuing this AD after determining that the unsafe condition described previously is likely to exist or develop on other products of the same type design.

### **AD Requirements**

This AD retains all requirements of AD 2025-13-12. This AD requires accomplishing the actions specified in EASA AD 2025-0197R1 described previously, except for any differences identified as exceptions in the regulatory text of this AD.

### **Explanation of Required Compliance Information**

In the FAA's ongoing efforts to improve the efficiency of the AD process, the FAA developed a process to use some civil aviation authority (CAA) ADs as the primary source of information for compliance with requirements for corresponding FAA ADs. The FAA has been coordinating this process with manufacturers and CAAs. As a result, EASA AD 2025-0197R1 is incorporated by reference in this AD. This AD requires compliance with EASA AD 2025-0197R1 in its entirety through that incorporation, except for any differences identified as exceptions in the regulatory text of this AD. Using common terms that are the same as the heading of a particular section in EASA AD 2025-0197R1 does not mean that operators need comply only with that section. For example, where the AD requirement refers to "all required actions and compliance times," compliance with this AD requirement is not limited to the section titled "Required

Action(s) and Compliance Time(s)” in EASA AD 2025-0197R1. Material required by EASA AD 2025-0197R1 for compliance will be available at regulations.gov under Docket No. FAA-2025-5390 after this AD is published.

### **Justification for Immediate Adoption and Determination of the Effective Date**

Section 553(b) of the Administrative Procedure Act (APA) (5 U.S.C. 551 *et seq.*) authorizes agencies to dispense with notice and comment procedures for rules when the agency, for “good cause,” finds that those procedures are “impracticable, unnecessary, or contrary to the public interest.” Under this section, an agency, upon finding good cause, may issue a final rule without providing notice and seeking comment prior to issuance. Further, section 553(d) of the APA authorizes agencies to make rules effective in less than thirty days, upon a finding of good cause.

An unsafe condition exists that requires the immediate adoption of this AD without providing an opportunity for public comments prior to adoption. The FAA has found that the risk to the flying public justifies forgoing notice and comment prior to adoption of this rule because an aileron FCRM electronic card exposed to hydraulic fluid has led to loss of control of an outboard aileron surface, and elevator and rudder FCRM electronic cards exposed to hydraulic fluid are also subject to failure due to their similarity of design. Control of the airplane is significantly degraded through the loss of control of an elevator or rudder surface, resulting in loss of control of the airplane. To mitigate this unsafe condition, the FAA issued interim AD 2025-13-12 to require replacement of affected FCRMs. However, those replacements are not repetitive, and the modification must be accomplished to eliminate the unsafe condition. Additionally, the compliance time in this AD is shorter than the time necessary for the public to comment and for publication of the final rule. Accordingly, notice and opportunity for prior public comment are impracticable and contrary to the public interest pursuant to 5 U.S.C. 553(b).

In addition, the FAA finds that good cause exists pursuant to 5 U.S.C. 553(d) for making this amendment effective in less than 30 days, for the same reasons the FAA found good cause to forgo notice and comment.

**Regulatory Flexibility Act**

The requirements of the Regulatory Flexibility Act (RFA) do not apply when an agency finds good cause pursuant to 5 U.S.C. 553 to adopt a rule without prior notice and comment. Because the FAA has determined that it has good cause to adopt this rule without notice and comment, RFA analysis is not required.

**Costs of Compliance**

The FAA estimates that this AD affects 39 airplanes of U.S. registry. The FAA estimates the following costs to comply with this AD:

**Estimated costs for required actions**

<b>Action</b>	<b>Labor cost</b>	<b>Parts cost</b>	<b>Cost per product</b>	<b>Cost on U.S. operators</b>
Retained actions from AD 2025-13-12	12 work-hours X \$85 per hour = \$1,020	\$110,256*	Up to \$111,276	Up to \$4,339,764
New actions	14 work-hours X \$85 per hour = \$1,190	\$1,044	\$2,234	\$87,126

\* Parts cost if all four FCRMs are replaced.

The FAA estimates the following costs to do any necessary on-condition action that would be required based on the results of any required actions. The FAA has no way of determining the number of aircraft that might need this on-condition action:

**Estimated costs of on-condition actions**

<b>Labor cost</b>	<b>Parts cost</b>	<b>Cost per product</b>
3 work-hours X \$85 per hour = \$255	Up to \$27,564	Up to \$27,819

## **Authority for This Rulemaking**

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs, describes in more detail the scope of the Agency's authority.

The FAA is issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701: General requirements. Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

## **Regulatory Findings**

This AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

- (1) Is not a "significant regulatory action" under Executive Order 12866, and
- (2) Will not affect intrastate aviation in Alaska.

## **List of Subjects in 14 CFR Part 39**

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

## **The Amendment**

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

## **PART 39 - AIRWORTHINESS DIRECTIVES**

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

### **§ 39.13 [Amended]**

2. The FAA amends § 39.13 by:

a. Removing Airworthiness Directive 2025-13-12, Amendment 39-23078 (90 FR 30577, July 10, 2025); and

b. Adding the following new airworthiness directive:

**2025-25-12 Airbus SAS:** Amendment 39-23220; Docket No. FAA-2025-5390; Project Identifier MCAI-2025-01470-T.

#### **(a) Effective Date**

This airworthiness directive (AD) is effective [INSERT DATE 15 DAYS AFTER DATE OF PUBLICATION IN THE FEDERAL REGISTER].

#### **(b) Affected ADs**

This AD replaces AD 2025-13-12, Amendment 39-23078 (90 FR 30577, July 10, 2025) (AD 2025-13-12).

#### **(c) Applicability**

This AD applies to all Airbus SAS Model A350-941 and -1041 airplanes, certificated in any category.

#### **(d) Subject**

Air Transport Association (ATA) of America Code 27, Flight Controls.

#### **(e) Reason**

This AD was prompted by a report of loss of control of an outboard aileron surface due to hydraulic fluid contaminating an electronic card of the flight control remote module (FCRM). The FAA is issuing this AD to address FCRM electronic cards

exposed to hydraulic fluid contamination. This condition, if not detected and corrected, could lead to runaway of rudder or elevator surface, resulting in loss of control of the airplane.

**(f) Compliance**

Comply with this AD within the compliance times specified, unless already done.

**(g) Requirements**

Except as specified in paragraph (h) of this AD: Comply with all required actions and compliance times specified in, and in accordance with, European Union Aviation Safety Agency (EASA) AD 2025-0197R1, dated September 26, 2025 (EASA AD 2025-0197R1).

**(h) Exceptions to EASA AD 2025-0197R1**

(1) Where EASA AD 2025-0197R1 refers to “07 May 2025 [the effective date of EASA AD 2025-0099],” this AD requires using July 25, 2025 (the effective date of AD 2025-13-12).

(2) Where EASA AD 2025-0197R1 refers to “12 June 2025 [the effective date of EASA AD 2025-0129],” this AD requires using July 25, 2025 (the effective date of AD 2025-13-12).

(3) Where EASA AD 2025-0197R1 refers to its effective date, this AD requires using the effective date of this AD.

(4) Where paragraph (1) of EASA AD 2025-0197R1 states “do not swap”, this AD requires replacing that text with “do not replace”.

(5) Where paragraph (2) of EASA AD 2025-0197R1 states “is reported on an aeroplane”, this AD requires replacing that text with “is documented in the aircraft maintenance records or has been reported to Airbus”.

(6) Where table 1 of EASA AD 2025-0197R1 states “close to”, this AD requires replacing that text with “adjacent to”.

(7) Where paragraph (13) of EASA AD 2025-0197R1 specifies installing certain software standards on the airplane in accordance with Airbus approved instructions, for this AD, those instructions must be approved using a method specified in paragraph (j)(2) of this AD.

(8) This AD does not adopt the “Remarks” section of EASA AD 2025-0197R1.

**(i) No Reporting Requirement**

Although material referenced in EASA AD 2025-0197R1 specifies to submit certain information to the manufacturer, this AD does not include that requirement.

**(j) Additional AD Provisions**

The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs)*: The Manager, AIR-520, Continued Operational Safety Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or responsible Flight Standards Office, as appropriate. If sending information directly to the manager of the Continued Operational Safety Branch, send it to the attention of the person identified in paragraph (k) of this AD and email to: AMOC@faa.gov.

(i) Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the responsible Flight Standards Office.

(ii) AMOCs approved previously for AD 2025-13-12 are approved as AMOCs for the corresponding provisions of EASA AD 2025-0197R1 that are required by paragraph (g) of this AD.

(2) *Contacting the Manufacturer*: For any requirement in this AD to obtain instructions from a manufacturer, the instructions must be accomplished using a method approved by the Manager, AIR-520, Continued Operational Safety Branch, FAA; or

EASA; or Airbus SAS's EASA Design Organization Approval (DOA). If approved by the DOA, the approval must include the DOA-authorized signature.

(3) *Required for Compliance (RC)*: Except as required by paragraph (j)(2) of this AD, where the service bulletins referenced in EASA AD 2025-0197R1 contain procedures or tests that are identified as RC, those procedures and tests must be done to comply with this AD; and where the alert operators transmission (AOT) referenced in EASA AD 2025-0197R1 contains paragraphs that are labeled as RC, the instructions in RC paragraphs, including subparagraphs under and RC paragraph, must be done to comply with this AD. Any procedures or tests, and instructions in paragraphs including subparagraphs under those paragraphs, as applicable, that are not identified as RC are recommended. Procedures and tests, and instructions in paragraphs including subparagraphs under those paragraphs, as applicable, that are not identified as RC may be deviated from using accepted methods in accordance with the operator's maintenance or inspection program without obtaining approval of an AMOC, provided the procedures, tests, and instructions, as applicable, identified as RC can be done and the airplane can be put back in an airworthy condition. Any substitutions or changes to procedures, tests, or instructions, as applicable, identified as RC require approval of an AMOC.

**(k) Additional Information**

For more information about this AD, contact Frank Carreras, Aviation Safety Engineer, FAA, 2200 South 216th St., Des Moines, WA 98198; phone: 206-231-3539; email: Frank.Carreras@faa.gov.

**(l) Material Incorporated by Reference**

(1) The Director of the Federal Register approved the incorporation by reference of the material listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this material as applicable to do the actions required by this AD, unless this AD specifies otherwise.

(i) European Aviation Safety Agency (EASA) AD 2025-0197R1, dated September 26, 2025.

(ii) [Reserved]

(3) For EASA material identified in this AD, contact EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; telephone +49 221 8999 000; email [ADs@easa.europa.eu](mailto:ADs@easa.europa.eu). You may find this material on the EASA website at [ad.easa.europa.eu](http://ad.easa.europa.eu).

(4) You may view this material at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206-231-3195.

(5) You may view this material at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, visit [www.archives.gov/federal-register/cfr/ibr-locations](http://www.archives.gov/federal-register/cfr/ibr-locations) or email [fr.inspection@nara.gov](mailto:fr.inspection@nara.gov).

Issued on December 19, 2025.

Peter A. White,  
Deputy Director, Integrated Certificate Management Division,  
Aircraft Certification Service.  
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