DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 31

[Docket No. FAA-2016-5424; Notice No. 31-16-01-SC]

Special Conditions: Ultramagic, S.A., Mark-32 Burner Series.

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of proposed special conditions.

SUMMARY: This action proposes special conditions for the Ultramagic, S.A., balloon models F-18, H-56, H-65, H-77, M-56, M-56C, M-65, M-65C, M-77, M-77C, M-90, M-105, M-120, M-130, M-145, M-160, N-180, N-210, N-250, N-300, N-355, N-425, S-70, S-90, S-105, S-130, S-160, T-150, T-180, T-210, V-56, V-65, V-77, V-90, V-105, and Z-90. These models will have a novel or unusual design feature associated with having the new Mark-32 Burner series. The applicable airworthiness regulations do not contain adequate or appropriate safety standards for this design feature. These proposed special conditions contain the additional safety standards that the Administrator considers necessary to establish a level of safety equivalent to that established by the existing airworthiness standards.

DATES: Send your comments on or before [insert a date 30 days after date of publication in the Federal Register].
ADDRESSES: Send comments identified by docket number FAA-2016-5424 using any of the following methods:

☐ Federal eRegulations Portal: Go to http://www.regulations.gov and follow the online instructions for sending your comments electronically.

☐ Mail: Send comments to Docket Operations, M-30, U.S. Department of Transportation (DOT), 1200 New Jersey Avenue, SE., Room W12-140, West Building Ground Floor, Washington, D.C., 20590-0001.

☐ Hand Delivery of Courier: Take comments to Docket Operations in Room W12-140 of the West Building Ground Floor at 1200 New Jersey Avenue, S.E., Washington, D.C., between 9 a.m., and 5 p.m., Monday through Friday, except Federal holidays.

☐ Fax: Fax comments to Docket Operations at 202-493-2251.

Privacy: The FAA will post all comments it receives, without change, to http://regulations.gov, including any personal information the commenter provides. Using the search function of the docket web site, anyone can find and read the electronic form of all comments received into any FAA docket, including the name of the individual sending the comment (or signing the comment for an association, business, labor union, etc.). DOT’s complete Privacy Act Statement can be found in the Federal Register published on April 11, 2000 (65 FR 19477-19478), as well as at http://DocketsInfo.dot.gov.

Docket: Background documents or comments received may be read at http://www.regulations.gov at any time. Follow the online instructions for accessing the docket or go to the Docket Operations in Room W12-140 of the West Building Ground Floor at 1200
New Jersey Avenue, SE., Washington, D.C., between 9 a.m., and 5 p.m., Monday through Friday, except Federal holidays.

FOR FURTHER INFORMATION CONTACT: John VanHoudt, FAA, Program and Procedures Branch, ACE-114, Small Airplane Directorate, Aircraft Certification Service, 901 Locust; Kansas City, Missouri 64106; telephone (816) 329-4142; facsimile (816) 329-4090.

SUPPLEMENTARY INFORMATION:

Comments Invited

We invite interested people to take part in this rulemaking by sending written comments, data, or views. The most helpful comments reference a specific portion of the special conditions, explain the reason for any recommended change, and include supporting data. We ask that you send us two copies of written comments.

We will consider all comments we receive on or before the closing date for comments. We will consider comments filed late if it is possible to do so without incurring expense or delay. We may change these special conditions based on the comments we receive.

Background

The MK-32 burner does introduce a particular novel aspect in terms of operation and performance—the primary modification being an oxygen augmented igniter system.

**Type Certification Basis**

Under the provisions of § 21.101, Ultramagic must show that the balloon models F-18, H-56, H-65, H-77, M-56, M-56C, M-65, M-65C, M-77, M-77C, M-90, M-105, M-120, M-130, M-145, M-160, N-180, N-210, N-250, N-300, N-355, N-425, S-70, S-90, S-105, S-130, S-160, T-150, T-180, T-210, V-56, V-65, V-77, V-90, V-105, and Z-90, as changed, continues to meet the applicable provisions incorporated by reference in Type Certificate No. B02CE or the applicable regulations in effect on the date of application for the change. The regulations incorporated by reference in the type certificate are commonly referred to as the "original type certification basis."

The Direccion General de Aviacion Civil originally type certificated this aircraft under its type certificate Numbers 3, 4, 18, 61, 147, and 247. The FAA validated these products under U.S. Type Certificate Number B02CE. On September 28, 2003, EASA began oversight of this product on behalf of Spain. The regulations incorporated by reference in B02CE are as follows:


b. 14 CFR part 31, effective on January 1990, as amended by 31-1 through 31-5 inclusive.


c. Equivalent level of Safety findings per provision of 14 CFR 21.21(b)(1):

(1) ACE-08-15\(^1\), August 1, 2008, Burners, 14 CFR 31.47(d).

(2) ACE-08-15A\(^2\), November 05, 2013, Burners, 14 CFR 31.47(d), for Model S-70.

If the Administrator finds that the applicable airworthiness regulations (i.e., 14 CFR part 31) do not contain adequate or appropriate safety standards for balloon models F-18, H-56, H-65, H-77, M-56, M-56C, M-65, M-65C, M-77, M-77C, M-90, M-105, M-120, M-130, M-145, M-160, N-180, N-210, N-250, N-300, N-355, N-425, S-70, S-90, S-105, S-130, S-160, T-150, T-180, T-210, V-56, V-65, V-77, V-90, V-105, and Z-90 because of a novel or unusual design feature, special conditions are prescribed under the provisions of § 21.16.

Special conditions are initially applicable to the model for which they are issued. Should the type certificate for that model be amended later to include any other model that incorporates the same or similar novel or unusual design feature, or should any other model already included on the same type certificate be modified to incorporate the same or similar novel or unusual design feature, the special conditions would also apply to the other model under § 21.101.

The FAA issues special conditions, as defined in 14 CFR 11.19, in accordance with § 11.38, and they become part of the type-certification basis under § 21.101.

**Novel or Unusual Design Features**


The oxygen augmentation and hydraulic control.

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Discussion

Based on the provisions of §§ 21.17 and 21.29 and the U.S.-EASA Technical Implementation Procedures for Airworthiness and Environmental Certification Between the Federal Aviation Administration of the United States of America and the European Aviation Safety Agency of the European Union, the following airworthiness requirements are applicable to this project and will remain active for three years from the date of application and form the Certification Basis:

a. Part 31, amendment 7 (The certification basis complied with according to the Ultramagic part 31 compliance checklist.).

b. Equivalent Level of Safety (ELOS) Findings: The FAA notes that it has issued an equivalent level of safety findings per provision of 14 CFR 21.21(b)(1), specifically ACE-08-15\textsuperscript{3} on August 1, 2008, Burners, § 31.47(d) and then extended the ELOS as ACE-08-15A\textsuperscript{4} on November 05, 2013, Burners, § 31.47(d), for the Model S-70. This ELOS has not been applied to the MK-32 and therefore not applicable.

3. Special conditions: The FAA notes that Ultramagic elected to comply with certain provisions of CS-23, amendment 3, that apply to oxygen systems. These provisions are applicable because there is an oxygen augmented igniter system available for the MK-32 burner. The below 14 CFR regulations, except § 23.1445, are harmonized with their CS-23, amendment 3, counterpart regulations and form the basis of this special condition.


§ 23.1445, Oxygen distribution system, paragraphs (a) and (b) states the following:

(a) Except for flexible lines from oxygen outlets to the dispensing units, or where shown to be otherwise suitable to the installation, nonmetallic tubing must not be used for any oxygen line that is normally pressurized during flight.
(b) Non-metallic oxygen distribution lines must not be routed where they may be subjected to elevated temperatures, electrical arcing, and released flammable fluids that might result from any probable failure.

§ 23.1451, Fire protection for oxygen equipment, paragraphs (a), (b), and (c) states the following:

Oxygen equipment and lines must—
(a) Not be in any designated fire zone.
(b) Be protected from heat that may be generated in, or escaped from, any designated fire zone.
(c) Be installed so that escaping oxygen cannot cause ignition of grease, fluid, or vapour accumulations that are present in normal operation or that may result from the failure or malfunction of any other system.

§ 23.1453, Protection of oxygen equipment from rupture, paragraphs (a) and (b) states the following:

(a) Each element of the oxygen system must have sufficient strength to withstand the maximum pressure and temperature in combination with any externally applied loads arising from consideration of limit structural loads that may be acting on that part of the system.
(b) Oxygen pressure sources and the lines between the source and shutoff means must be:
   (1) Protected from unsafe temperatures; and
   (2) Located where the probability and hazard of rupture in a crash landing are minimized.

§ 23.1445 is the only significant regulatory difference, which states the following:

Part 23 requires crewmembers be able to reserve a minimum supply for themselves when they share a common source of O2 with passengers.
As the oxygen system is not utilized for breathing, this Significant Standard Difference (SSD) does not apply.

In addition, the FAA notes that Ultramagic offers an optional hydraulic kit. This kit is a hydraulic system that actuates the burners’ fuel valve. Since part 31 does not have provisions for hydraulic systems, § 23.1435, Hydraulic systems, will provide the basis for the hydraulic system special conditions contained herein. No SSD is associated with this regulation.

**Applicability**

As discussed above, these special conditions are applicable to the Model Numbers F-18, H-56, H-65, H-77, M-56, M-56C, M-65, M-65C, M-77, M-77C, M-90, M-105, M-120, M-130, M-145, M-160, N-180, N-210, N-250, N-300, N-355, N-425, S-70, S-90, S-105, S-130, S-160, T-150, T-180, T-210, V-56, V-65, V-77, V-90, V-105, and Z-90 balloons. Should Ultramagic, S.A. apply at a later date for a change to the type certificate to include another model incorporating the same novel or unusual design feature, the special conditions would apply to that model as well.

**Conclusion**

This action affects only certain novel or unusual design features on one model series of burners. It is not a rule of general applicability.

**List of Subjects in 14 CFR Part 31**

Aircraft, Aviation safety.

The authority citation for these special conditions is as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701-44702, 44704.
The Proposed Special Conditions


   (a) In addition to the provisions of part 31, amendment 7, the applicant must design the MK-32 Burner to comply with the requirements, as described below, with respect to the igniter oxygen augmentation system and hydraulic burner valve actuation system:

   **Oxygen distribution system**

   (1) Except for flexible lines from oxygen outlets to the dispensing units, or where shown to be otherwise suitable to the installation, nonmetallic tubing must not be used for any oxygen line that is normally pressurized during flight.

   (2) Nonmetallic oxygen distribution lines must not be routed where they may be subjected to elevated temperatures, electrical arcing, and released flammable fluids that might result from any probable failure.

   **Fire protection for oxygen equipment**

   Oxygen equipment and lines must:

   (1) Not be installed in any designated fire zones.

   (2) Be protected from heat that may be generated in, or escape from, any designated fire zone.
(3) Be installed so that escaping oxygen cannot come in contact with and cause ignition of grease, fluid, or vapor accumulations that are present in normal operation or that may result from the failure or malfunction of any other system.

Protection of oxygen equipment from rupture

(1) Each element of the oxygen system must have sufficient strength to withstand the maximum pressure and temperature, in combination with any externally applied loads arising from consideration of limit structural loads that may be acting on that part of the system.

(2) Oxygen pressure sources and the lines between the source and the shutoff means must be:

(i) Protected from unsafe temperatures; and

(ii) Located where the probability and hazard of rupture in a crash landing are minimized.

Hydraulic systems

(1) Design. Each hydraulic system must be designed as follows:

(i) Each hydraulic system and its elements must withstand, without yielding, the structural loads expected in addition to hydraulic loads.

(ii) A means to indicate the pressure in each hydraulic system which supplies two or more primary functions must be provided to the flight crew.

(iii) There must be means to ensure that the pressure, including transient (surge) pressure, in any part of the system will not exceed the safe limit above design operating pressure and to prevent excessive pressure resulting from fluid volumetric changes in all lines which are likely to remain closed long enough for such changes to occur.
(iv) The minimum design burst pressure must be 2.5 times the operating pressure.

(2) Tests. Each system must be substantiated by proof pressure tests. When proof tested, no part of any system may fail, malfunction, or experience a permanent set. The proof load of each system must be at least 1.5 times the maximum operating pressure of that system.

(3) Accumulators. A hydraulic accumulator or reservoir may be installed on the engine side of any firewall, if—

(i) It is an integral part of an engine or propeller system; or

(ii) The reservoir is nonpressurized and the total capacity of all such nonpressurized reservoirs is one quart or less.

(b) Ultramagic, through EASA, will provide the FAA with all Airworthiness Directives issued against the changed type design, if any, and a plan for resolving the unsafe conditions for the FAA type design.

Issued in Kansas City, Missouri, on March 28, 2016.

Mel Johnson
Acting Manager, Small Airplane Directorate
Aircraft Certification Service

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